

Interferometry on the Fringe: Cutting Edge Developments at Lowell

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Welcome to Acronym Word Salad

- ▶ BFT
- ▶ MoonLITE
- ▶ AeSI
- ▶ KISS

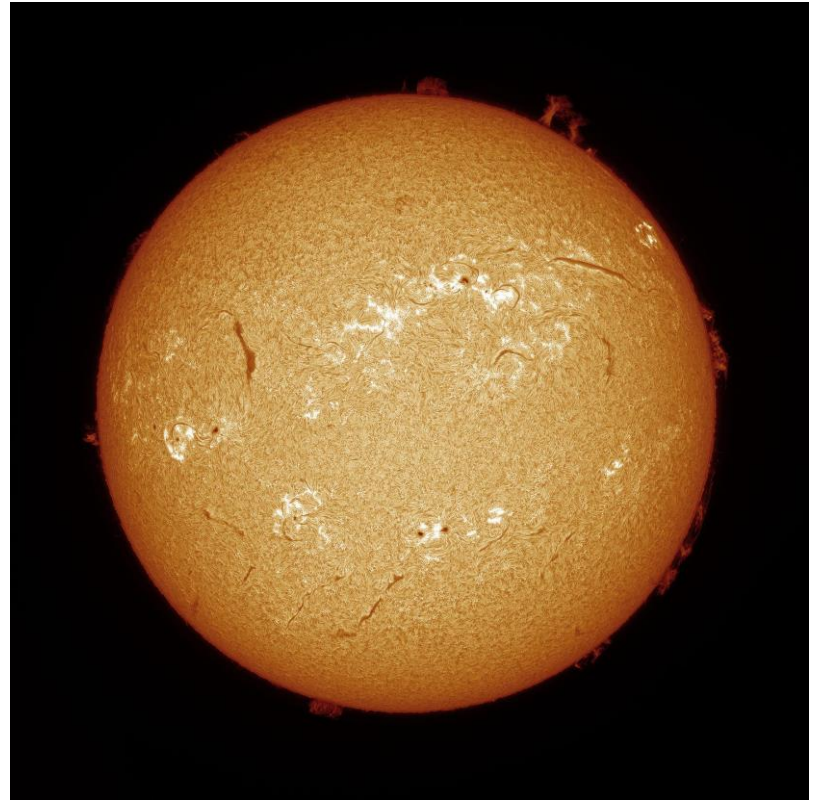


Missing *Affordable* Low-Hanging Fruit

- ▶ Bright, small objects
- ▶ Imaging with dense $\{u,v\}$ coverage
- ▶ Aggregate **30 years** of 'lessons learned' in design, construction, operations, and data reduction

Primary Target:

Main Sequence STARS



BFT – Big Fringe Telescope: Overview

- ▶ **Compelling, *focused* science case**
 - ▶ Exoplanet host surface imaging
 - ▶ Solar analog surface imaging
 - ▶ Exoplanet transit movies
 - ▶ Component-resolved binary stars
- ▶ **Additional general science cases**
- ▶ **Affordable architecture**
 - ▶ COTS parts, 30-years-newer technology, eliminate vacuum & other expensive infrastructure, true parts commonality
 - ▶ Construction & O&M costs baked into design from start
 - ▶ ***Single, robotic observational mode***
 - ▶ *Requirement for <3% annual operations costs*



Focused 'Marquee' Science Program

▶ EXPRES EPRV 100 Earths

- ▶ Brewer+ 2020, table 1
- ▶ 65 objects
- ▶ 0.40-0.60mas: 31/65
- ▶ 0.35-0.55mas: 31/65
- ▶ $V < 7.6$ and $K < 5.6$ for both

▶ Exoplanet transit movies

- ▶ TFOP: 29 targets with TESS
 $G < 7.5$, $R_{\text{planet}} > 5 \text{ RE}$, $DE > -10$,
median $\sim 0.40\text{mas}$

▶ Solar analogs

- ▶ S-cubed: 69 targets from Radick+ 2018
- ▶ 0.40-0.60mas: 35/69
- ▶ $V < 6.7$ and $K < 5.0$ for all

▶ Resolved binaries

- ▶ Pourbaix SB9 has 77 targets with $DE > -10$ and:
- ▶ $P < 1000\text{d}$, $a \sim 5\text{-}360\text{mas}$, $0.30\text{-}0.70\text{mas}$



Overall Requirements

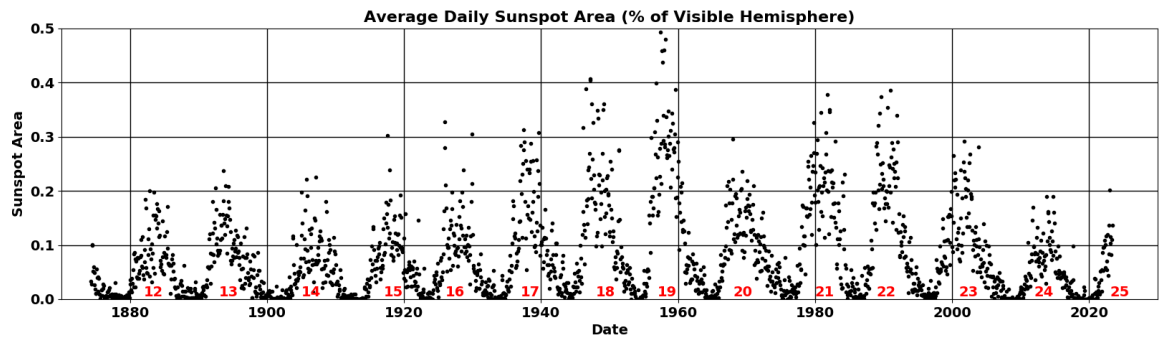
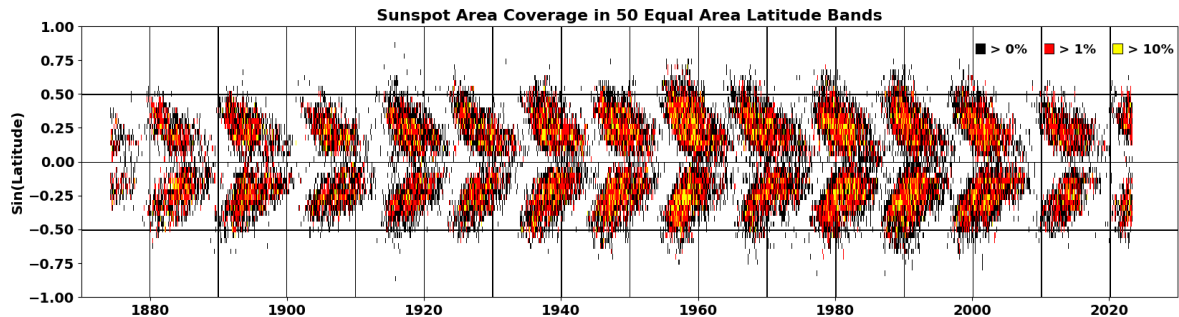
▶ **Resolution**

- ▶ Target 'sweet spot' of 0.40 – 0.60 mas, 20 μ as pixels
- ▶ Snapshot imaging

▶ **Sensitivity**

- ▶ Visible (Johnson V, R): < 7.6 (requirement) < 8.0 (goal)
- ▶ Near-infrared (Johnson H, K): < 5.6 (requirement) < 6.0 (goal)

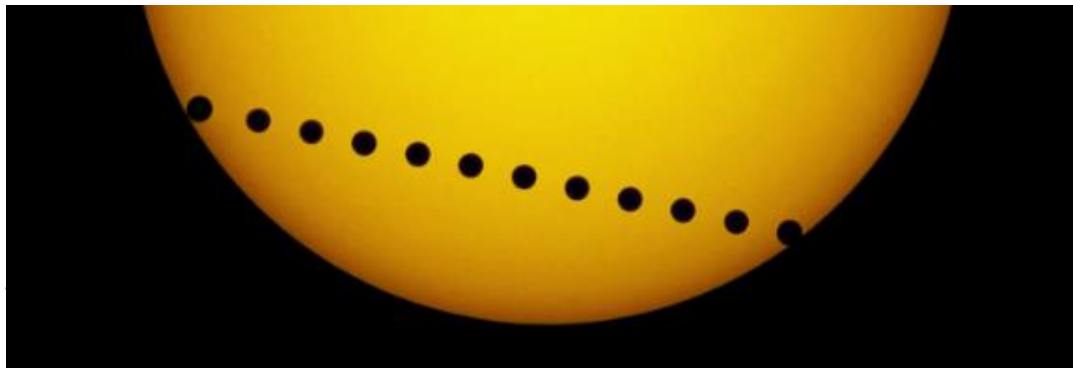




<http://SolarCycleScience.com>

2023/07 Hathaway

‘Butterfly’
diagrams



Exoplanet
transits

Required Resolution

▶ Number of pixels across star?

- ▶ Relative size of spots to disk
- ▶ Spot migration tracking
- ▶ **30 x 30 pixels is target**

- ▶ Imaging resolution = $1.22 \lambda/B$
 - ▶ 'Airy criterion'
- ▶ Modeling resolution = $0.25 \lambda/B$
 - ▶ 'Michelson criterion'

▶ Overall facility scale

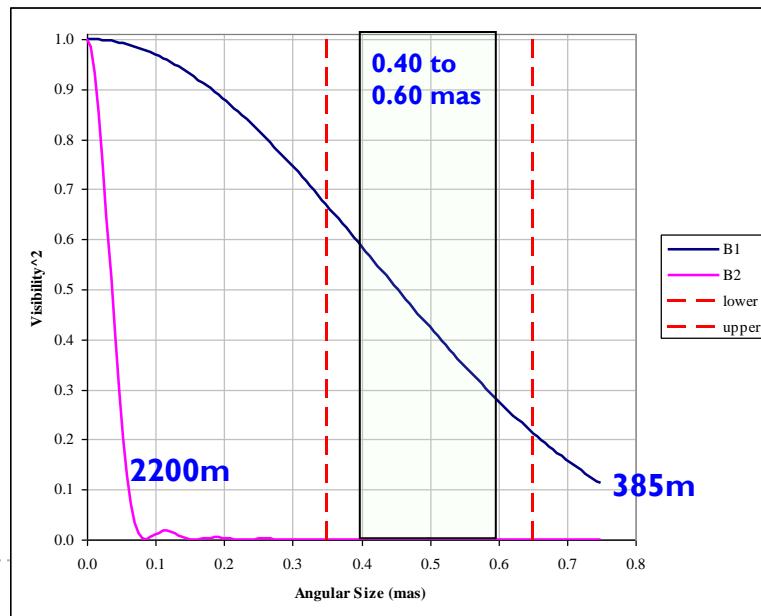
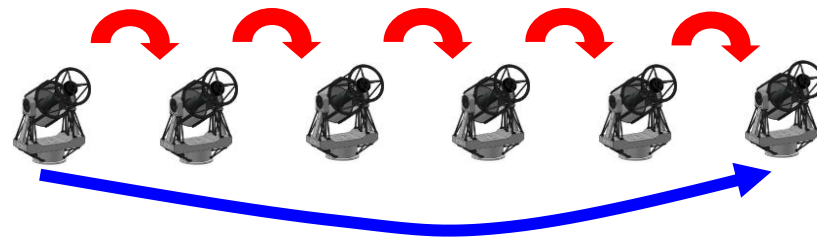
- ▶ 'Pixel' scale: dictates longest baseline
- ▶ Fringe tracking: dictates short baseline spacing
- ▶ Longest baseline has to be made up of short baselines

- ▶ **0.40-0.60 mas stars with 20uas pixels → 2.2km**



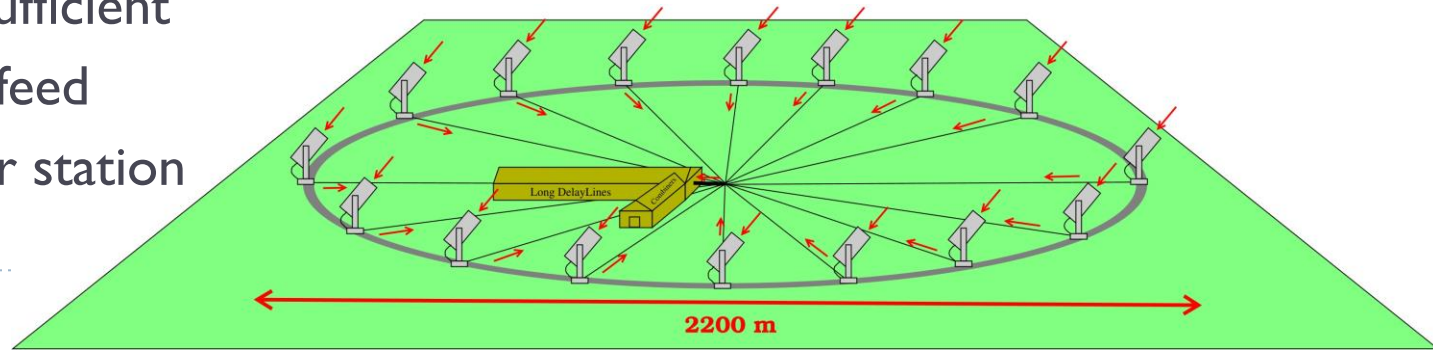
Wavelength-Baseline Bootstrapping

- ▶ Track at H-band with short baselines, short atmospheric coherence times
- ▶ Image in R-band with medium, long baselines, long synthetic coherence times
- ▶ H-band fringe tracking with $V^2 > 0.20$ means for 0.50 mas star, $B_{\min} = 385\text{m}$



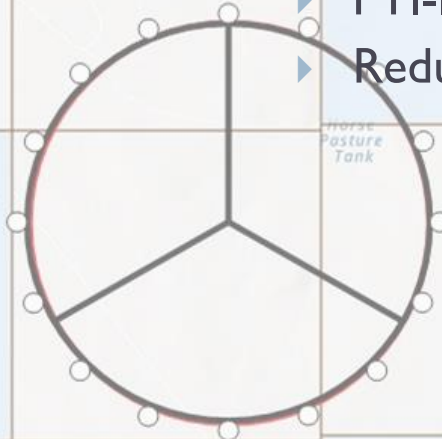
Nominal Layout

- ▶ 16 x 0.5 m telescopes
- ▶ 2,200m ring
 - ▶ Sufficient angular resolution for 0.40-0.60 mas targets, $\sim 30 \times 30$ 20 μas pixels
- ▶ Small telescopes
 - ▶ ‘Commodity item’
 - ▶ ‘Station 0’ robotic operations already started at Lowell
 - ▶ Enough area for sensitivity requirement
 - ▶ Tip-tilt is sufficient
 - ▶ Siderostat feed
 - ▶ $\sim \$150\text{k}$ per station



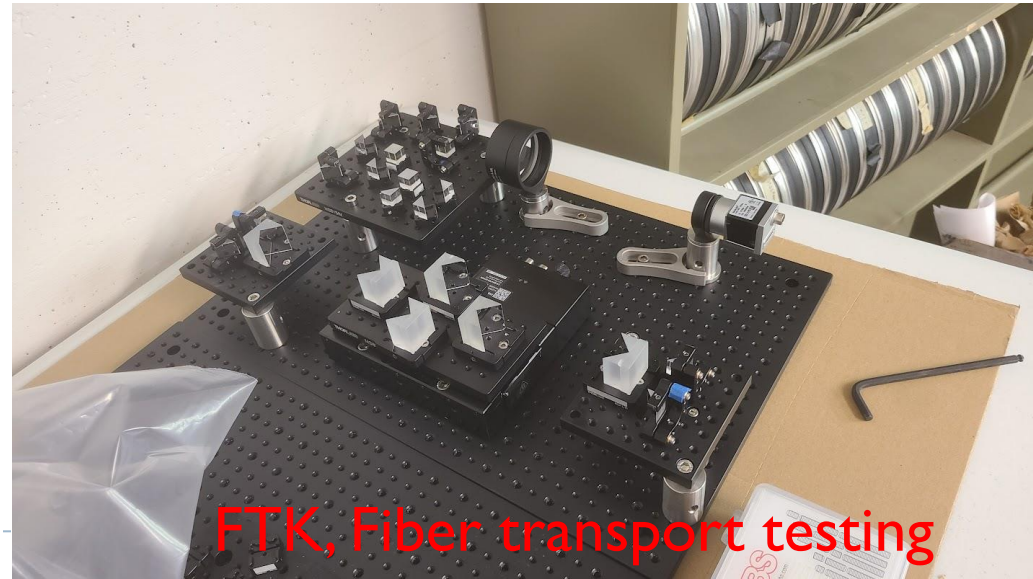
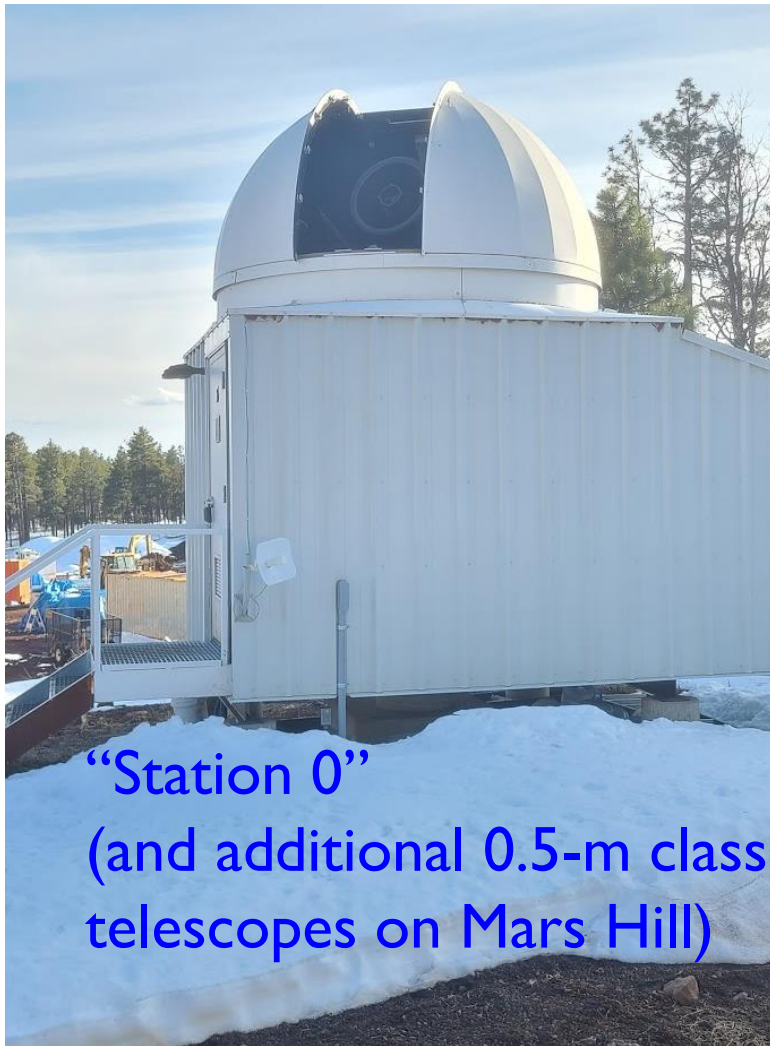
Nominal Infrastructure

- ▶ Lots of high, flat, available land in northern Arizona
- ▶ No vacuum systems
 - ▶ Eliminate significant cost infrastructure (both construction & O&M)
- ▶ Fiber beam transport
 - ▶ Testbed in lab at Lowell already
- ▶ Next-gen high-speed delay lines
 - ▶ Current DLs are 30+ year old technology
 - ▶ NSF ATI submitted for development
- ▶ Robotic operations
 - ▶ PTI-like automation
 - ▶ Reduce ops costs

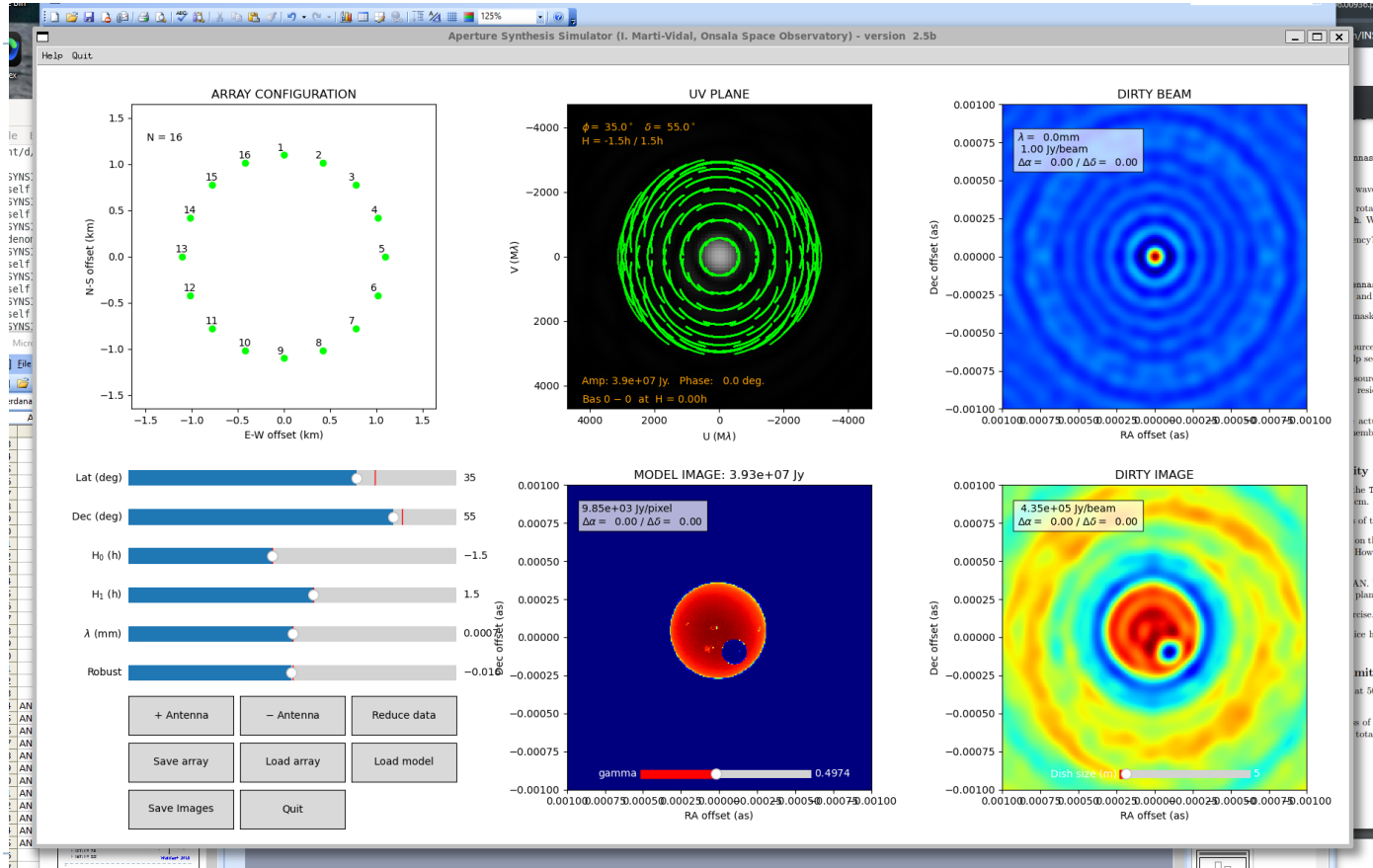


Scalable Development

- ▶ Lab → single baseline → full deployment



Snapshot from simulated planet transit movie



MoonLITE



What is MoonLITE?

- ▶ LITE = Lunar InTerferometry Explorer
- ▶ A **submitted** NASA Astrophysics Pioneers proposal
- ▶ A two-element, 100 meter Michelson interferometer
- ▶ CLPS-delivered to lunar surface
- ▶ Capable of $V=17$ **isolated objects**, for objects 0.1 - 1.6mas in size, measure 0.1-5.0% sizes



Lester (2006)

“The only thing the moon
has to offer astronomy is

dust

and **gravity**”

(slightly paraphrased)



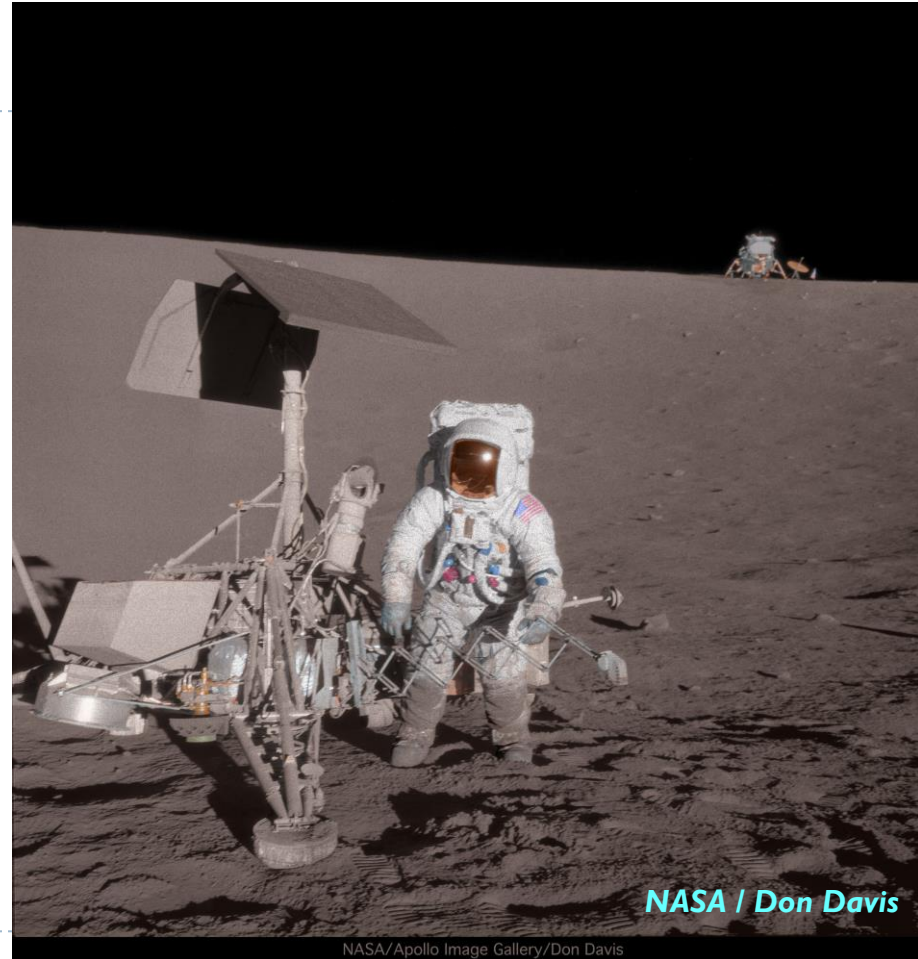
Dust: LUT on board Chang'e-3 Lander

- ▶ UV telescope with **years** of operations (2015 - 2018+)
- ▶ **Dust not a problem**



Gravity

- ▶ It's not a bug, it's a **feature**
 - ▶ Eg. Surveyor 3, Apollo 12: 180m baseline, stable relative position for the past 50 years
 - ▶ Nearly perfect for large optical interferometers
 - ▶ Formation flying is unsolved, expensive
- ▶ Greatly simplifies pointing
 - ▶ Stable reaction mass
 - ▶ 'Solved' for orbital platforms, but expensive and buggy – Eg. HST reaction wheels, Kepler, IUE, etc.



NASA / Don Davis

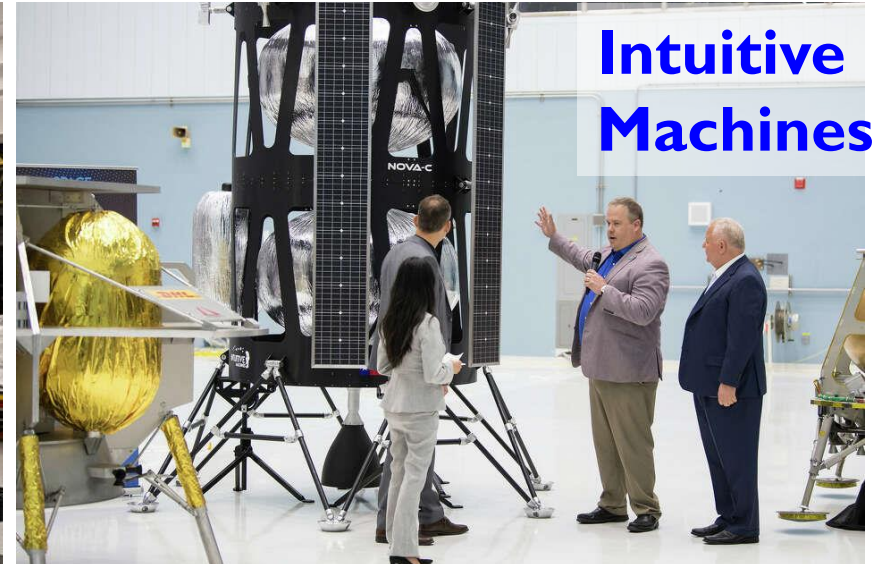
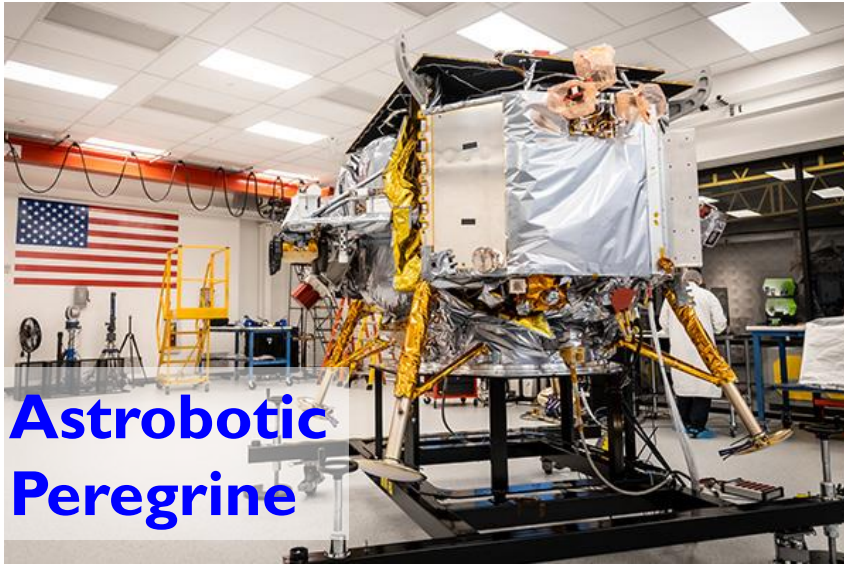
NASA/Apollo Image Gallery/Don Davis

Other Features

- ▶ **No atmosphere**
 - ▶ No atmospheric coherence time limit
 - ▶ A 2" aperture has greater sensitivity than an 8m VLTI aperture after first second of integration; 300+ sec possible
 - ▶ Free vacuum → clean beam propagation, no vacuum machinery
- ▶ **Stable surface**
 - ▶ Apollo seismometer data indicates <20nm vibrational background on week+ timescales



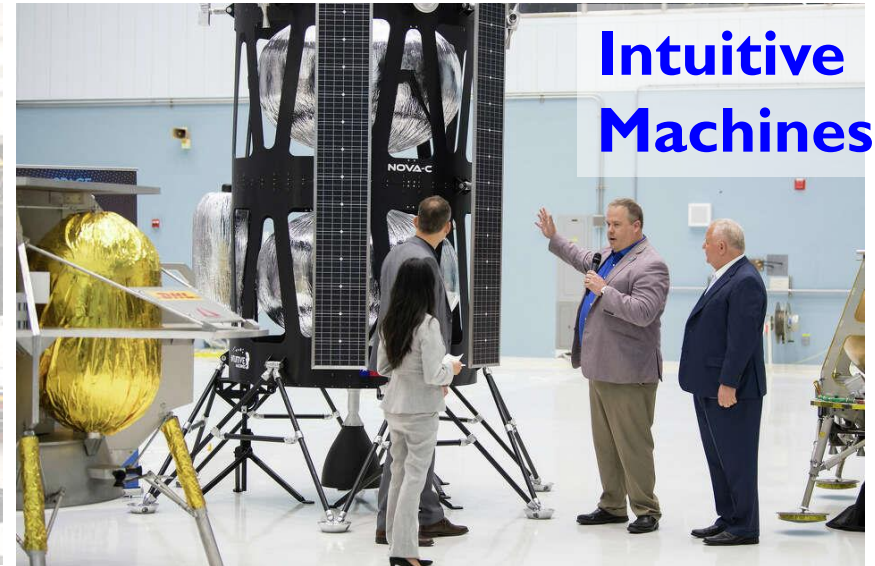
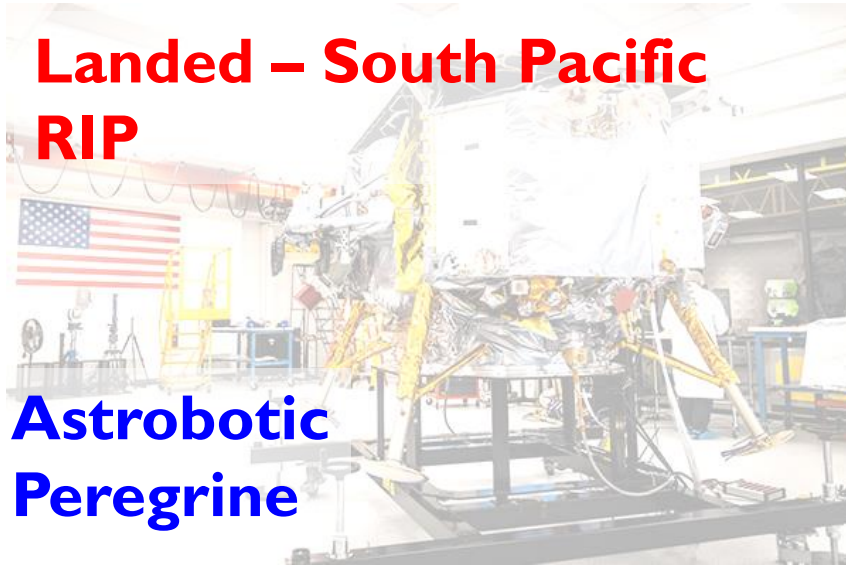
NASA Commercial Lunar Payload Services (CLPS) Landers



- ▶ Hosted payloads to the lunar surface, with rover
- ▶ **Allowable under NASA Astrophysics Pioneers**



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NASA Commercial Lunar Payload Services (CLPS) Landers

Launched Feb 15

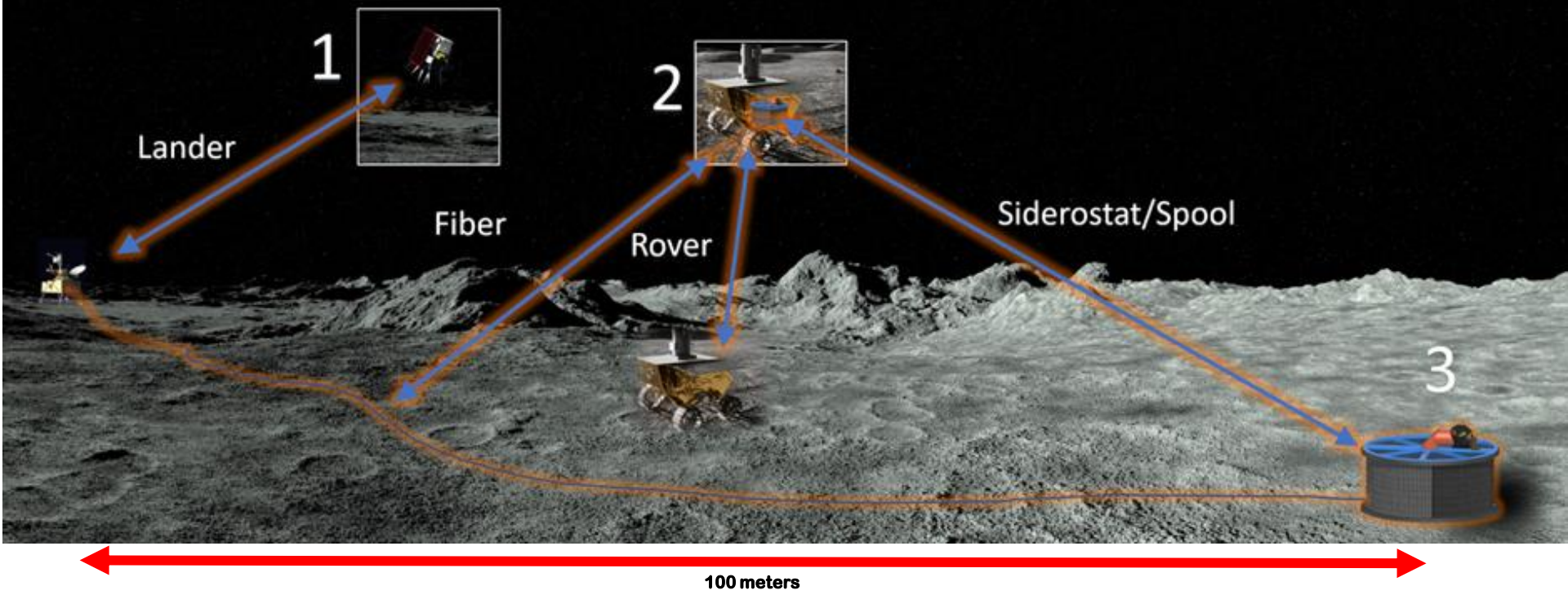
Landed on Feb 22



The Pioneers CLPS Box

- ▶ \$20M cap
- ▶ 50 kg, 200 W, 300kbps
- ▶ Daytime operations *only*





Emphasis on simplicity: one deployment step

Given lunar surface stability, hosting by lander, resources can be focused on the experiment itself



MoonLITE: Lunar InTerferometry Explorer

collector #1 (w/fiber relay)
beam combiner instrument

CLPS-
provided
lander
(power, comms)

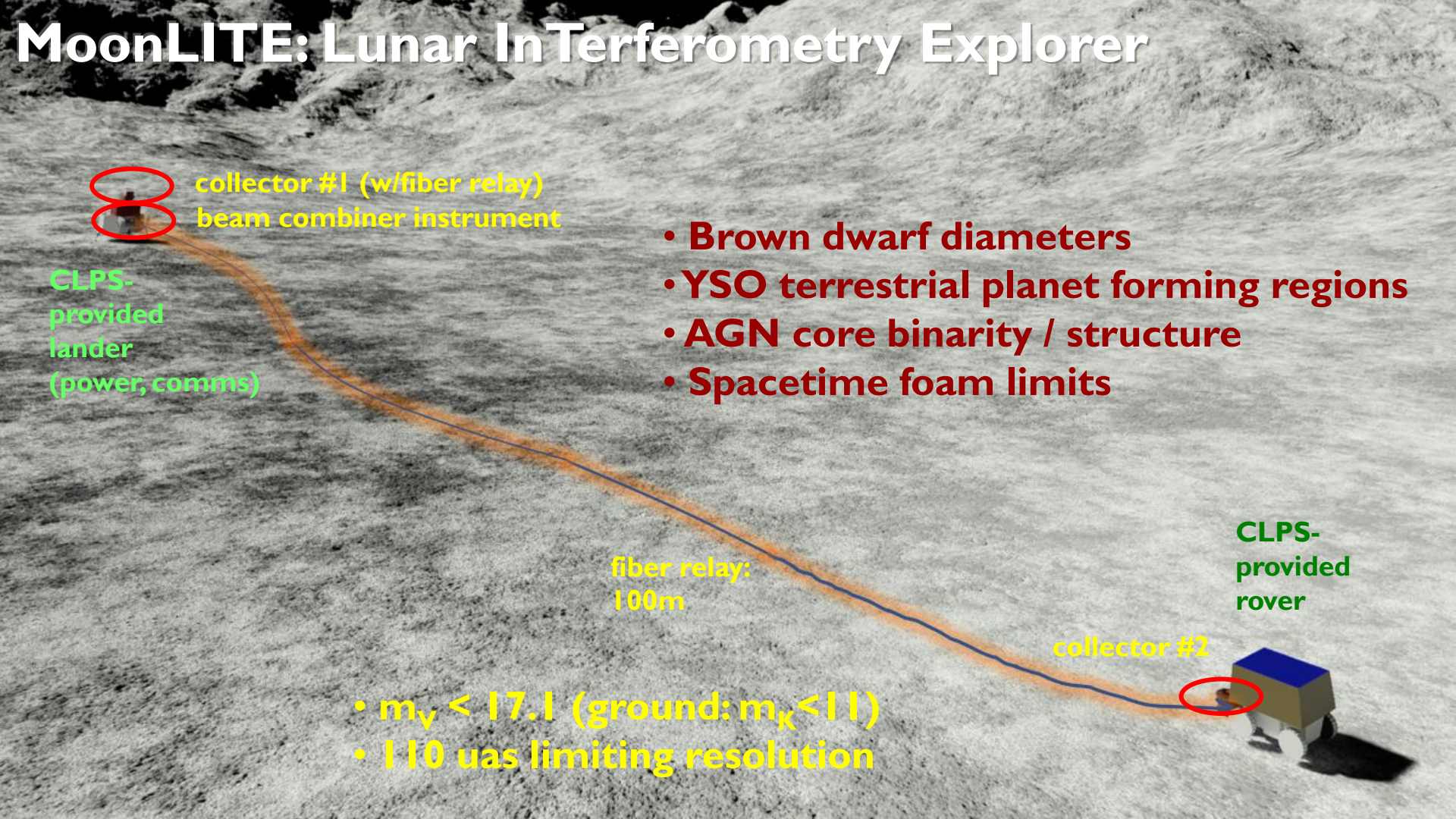
- **Brown dwarf diameters**
- **YSO terrestrial planet forming regions**
- **AGN core binarity / structure**
- **Spacetime foam limits**

fiber relay:
100m

CLPS-
provided
rover

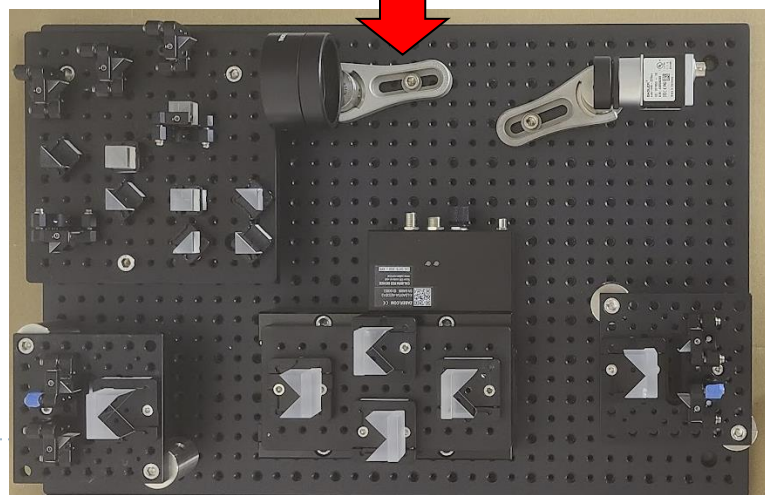
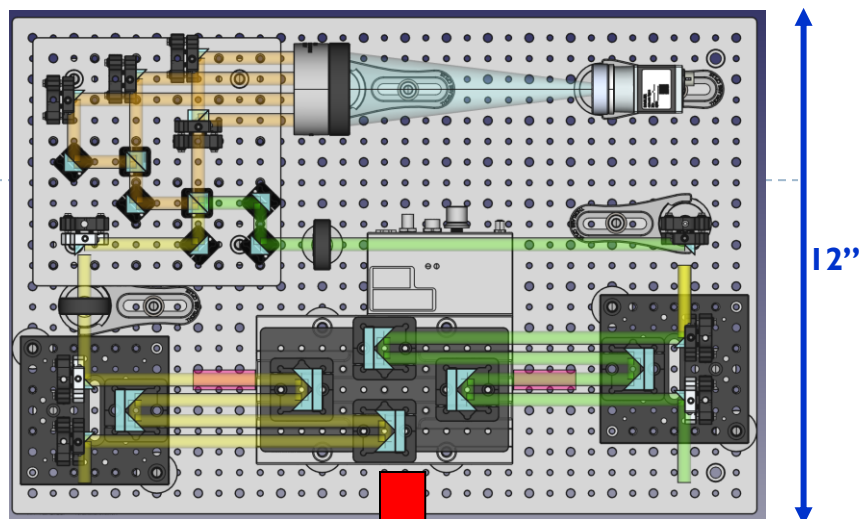
collector #2

- $m_v < 17.1$ (ground: $m_k < 11$)
- 110 μ s limiting resolution



Status

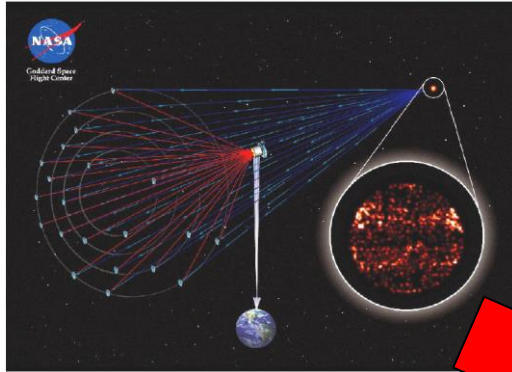
- ▶ Proposal declined
 - ▶ **Resubmission encouraged**
 - ▶ Supporting NASA SAT, NSF ATI proposals also submitted
- ▶ Progress with building opEDU with internal funds
- ▶ Aim to test opEDU in lab and then *on-sky* with telescopes available at Lowell
 - ▶ Tech can apply to BFT as well



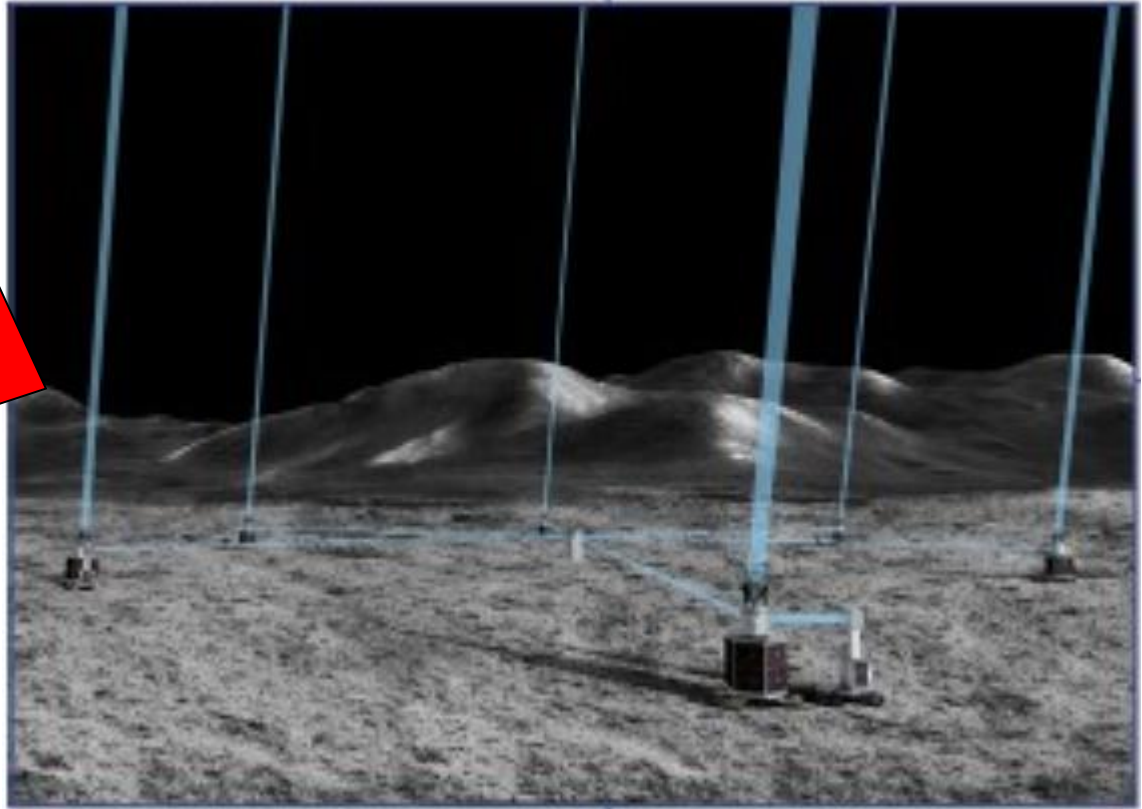
AeSI



Artemis-enabled Stellar Imager



- ▶ NASA NIAC-funded **study**
- ▶ 30 x 1m telescopes
- ▶ 500m @ 1200Å



Keck Institute of Space Studies (KISS) Study Program



KISS Study Program: “Astronomical Optical Interferometry from the Lunar Surface”



2023 Study Programs

Blazing Paths to Observing Stellar and Exoplanet Particle Environments



Determining the Interior Structure of Uranus



The Biology of Biosignature Detection



Managing Soil Organic Carbon for Climate Change Mitigation

